

Short Course on Experimental Dynamic Substructuring

Module #2: General theory



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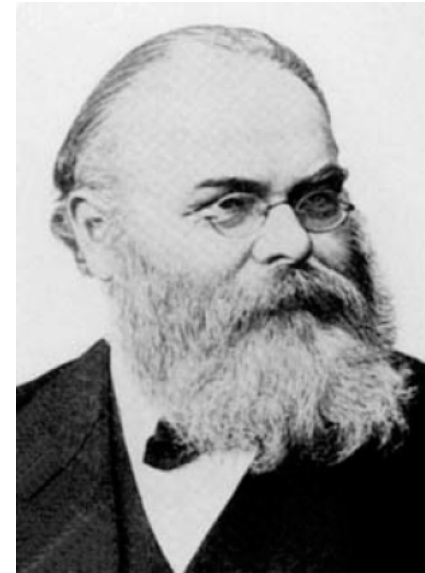
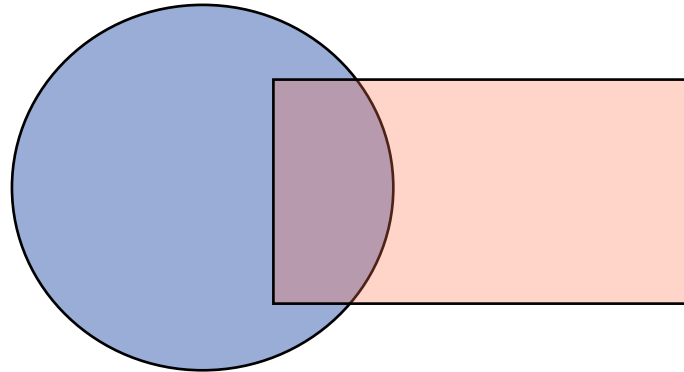
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Short Course Notes For:

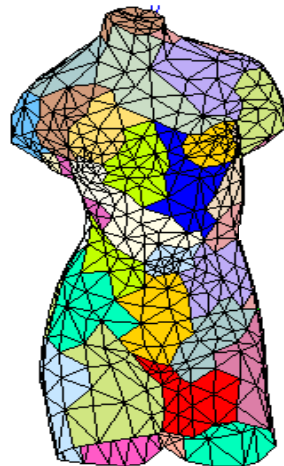
January 23, 2016, IMAC, Orlando, Florida

Some history – Divide and Conquer

- The first idea of decoupling and re-assembling to solve a problem was by H. Schwarz, in 1890, who wanted to prove the existence and uniqueness of the solution to the Laplace equation in a complex domain:



- Later, with the work of Courant (1943), the idea of subdivision was used to define Ritz approximation per elements. Finite Elements were born ...

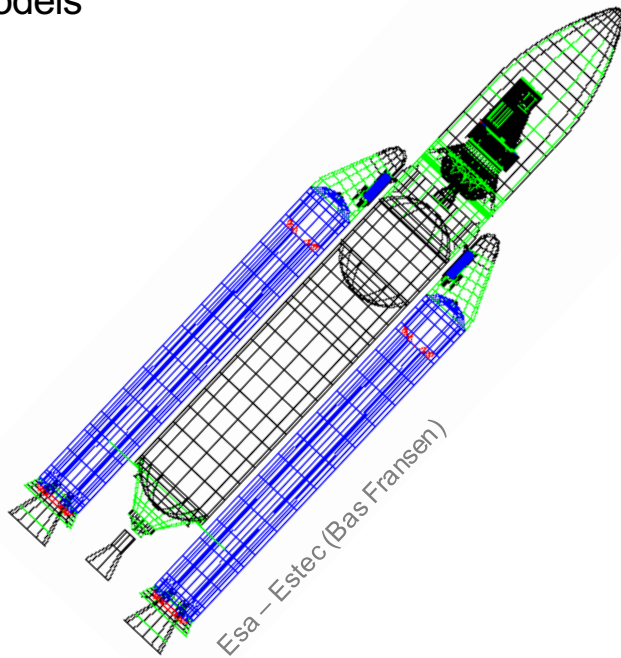


P. Gosselet – ENS Cachan

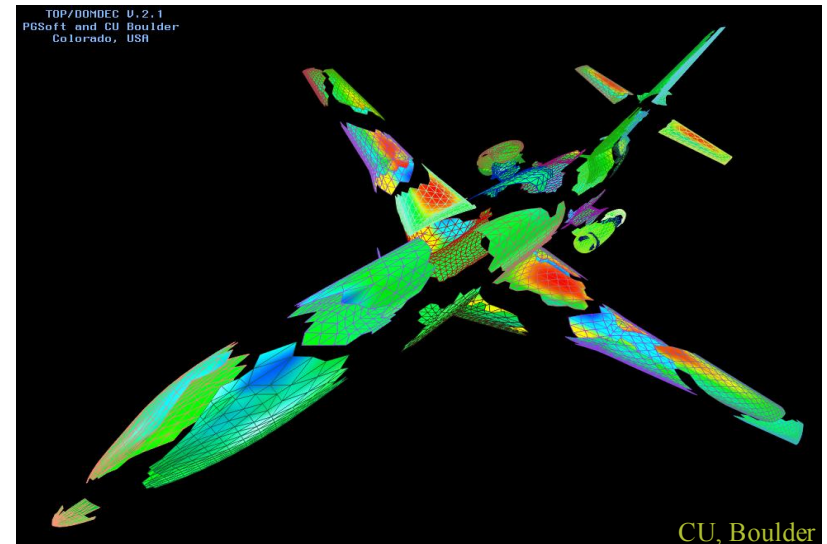


Some history – Divide and Conquer

- In the 60's the idea of partitioning a finite element model in substructures was proposed to reduce the complexity of the models



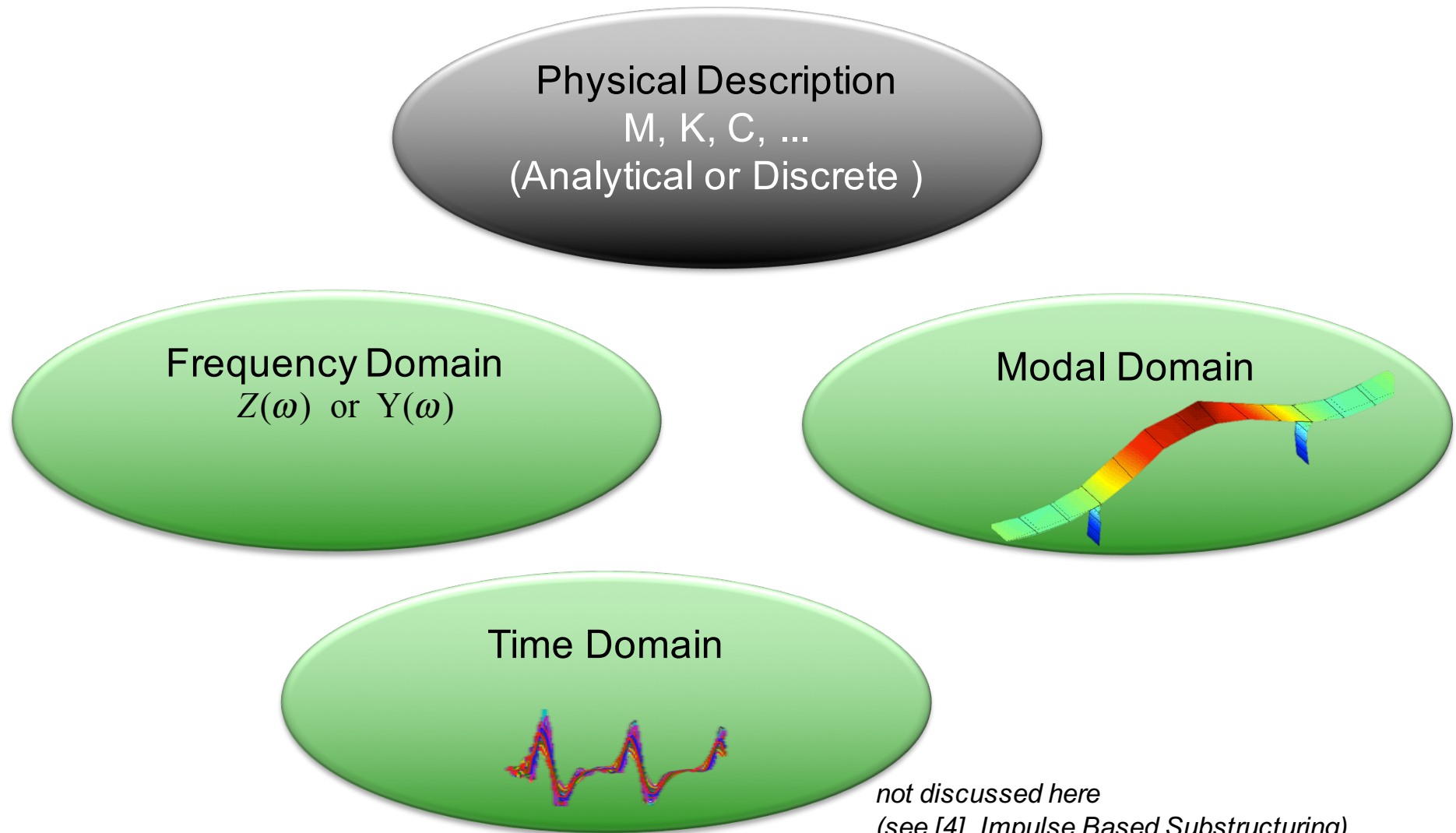
- In the 80's, in order to speed-up the solution of linear algebraic systems, like in $Ku=F$, the computational domain was cut in sub-domains in order to share work amongst several CPUs (parallel computing). That was the start of *Domain Decomposition*.



- End of the 60's, but especially in the 80's, decomposing a structure in experimental mechanics was applied in order to simplify the testing. Today new interest thanks to new understanding of methods, better sensors and acquisition.

Different Domains to apply substructuring [1]

Depending on how the dynamics of the components are described



Outline

- **Coupling the physical model**
 - ❑ 3-field form
 - ❑ Primal assembly
 - ❑ Dual assembly
 - ❑ Summary & Remarks
- **Coupling in the frequency domain**
 - ❑ Assembled admittance
 - ❑ Example
 - ❑ Summary & Remarks
- **Dual coupling in representation space**
- **Weak interface compatibility**
 - ❑ By projection
 - ❑ By filtering
- **Summary**

References and bibliography

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Coupling the physical model

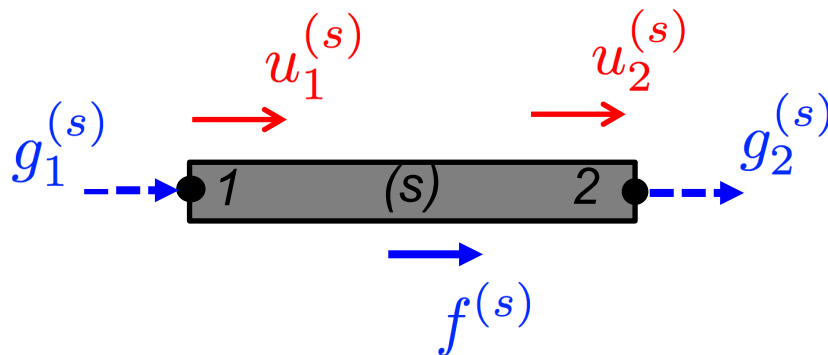
- Dynamic equation of one substructure (s)

*link forces on interface
(to be determined from global dynamics)*

$$\mathbf{M}^{(s)} \ddot{\mathbf{u}}^{(s)}(t) + \mathbf{C} \dot{\mathbf{u}}(t) + \mathbf{K}^{(s)} \mathbf{u}^{(s)}(t) = \mathbf{f}^{(s)}(t) + \mathbf{g}^{(s)}(t)$$

externally applied forces

- Example of bars.



$$\mathbf{K}^{(s)} = k^{(s)} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

$$\mathbf{M}^{(s)} = m^{(s)} \begin{bmatrix} 1/2 & 0 \\ 0 & 1/2 \end{bmatrix}$$

$$\mathbf{u}^{(s)} = \begin{bmatrix} u_1^{(s)} \\ u_2^{(s)} \end{bmatrix}$$

$$\mathbf{g}^{(s)} = \begin{bmatrix} g_1^{(s)} \\ g_2^{(s)} \end{bmatrix}$$

Coupling the physical model

- **Block-matrix** notation for n substructures

$$\mathbf{M}^{(s)} \ddot{\mathbf{u}}^{(s)}(t) + \mathbf{C} \dot{\mathbf{u}}(t) + \mathbf{K}^{(s)} \mathbf{u}^{(s)}(t) = \mathbf{f}^{(s)}(t) + \mathbf{g}^{(s)}(t)$$

$$\boxed{\mathbf{M} \ddot{\mathbf{u}} + \mathbf{C} \dot{\mathbf{u}} + \mathbf{K} \mathbf{u} = \mathbf{f} + \mathbf{g}}$$

unassembled system

$$\mathbf{M} \triangleq \text{diag} \left(\mathbf{M}^{(1)}, \dots, \mathbf{M}^{(n)} \right) = \begin{bmatrix} \mathbf{M}^{(1)} & \cdot & \cdot \\ \cdot & \ddots & \cdot \\ \cdot & \cdot & \mathbf{M}^{(n)} \end{bmatrix}$$

$$\mathbf{C} \triangleq \text{diag} \left(\mathbf{C}^{(1)}, \dots, \mathbf{C}^{(n)} \right)$$

$$\mathbf{K} \triangleq \text{diag} \left(\mathbf{K}^{(1)}, \dots, \mathbf{K}^{(n)} \right)$$

$$\mathbf{u} \triangleq \begin{bmatrix} \mathbf{u}^{(1)} \\ \vdots \\ \mathbf{u}^{(n)} \end{bmatrix}, \quad \mathbf{f} \triangleq \begin{bmatrix} \mathbf{f}^{(1)} \\ \vdots \\ \mathbf{f}^{(n)} \end{bmatrix}, \quad \mathbf{g} \triangleq \begin{bmatrix} \mathbf{g}^{(1)} \\ \vdots \\ \mathbf{g}^{(n)} \end{bmatrix}$$

Coupling the physical model: 3-field form

- The system is said to be **assembled** when **2 interface conditions** are met:
 - Compatibility of the displacements on the interface

$$Bu = \mathbf{0}$$

 *signed Boolean matrix (0,1,-1)*

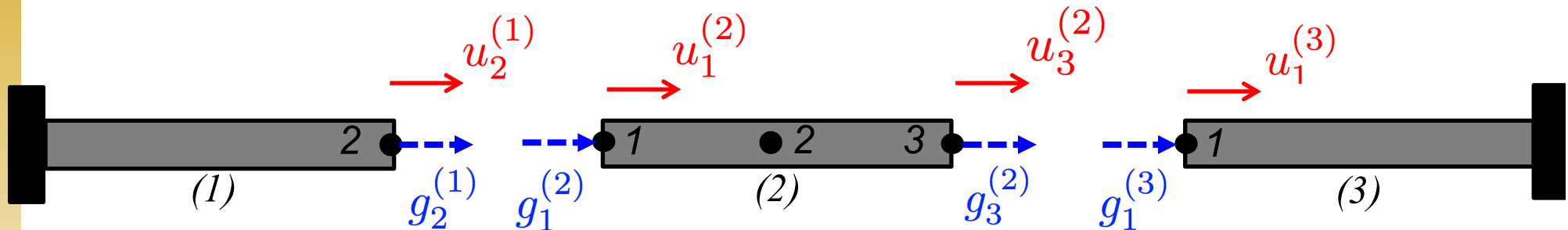
- Equilibrium of the interface forces (actio-reactio)

$$L^T g = \mathbf{0}$$

 *Boolean matrix with (0,1)*

Coupling the physical model: 3-field form

- Example of bars.



Compatibility on each interface

$$\left\{ \begin{array}{c} \text{\# interface connections} \\ \left[\begin{array}{ccccc} 1 & -1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & -1 \end{array} \right] \end{array} \right\} \begin{array}{c} \text{\# unassembled dofs} \\ \left[\begin{array}{c} u_2^{(1)} \\ u_1^{(2)} \\ u_2^{(2)} \\ u_3^{(2)} \\ u_1^{(3)} \end{array} \right] \end{array} = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$

$$Bu = \mathbf{0}$$

Computes jump on interface

Equilibrium of internal forces on each assembled dof

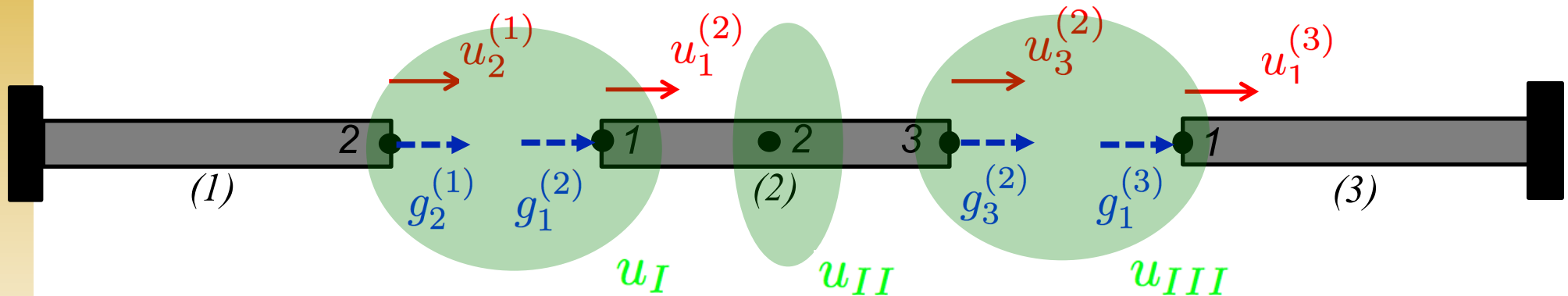
$$\left\{ \begin{array}{c} \text{\# assembled dof} \\ \left[\begin{array}{ccccc} 1 & 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 & 1 \end{array} \right] \end{array} \right\} \begin{array}{c} \text{\# unassembled dofs} \\ \left[\begin{array}{c} g_2^{(1)} \\ g_1^{(2)} \\ g_2^{(2)} \\ g_3^{(2)} \\ g_1^{(3)} \end{array} \right] \end{array} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$$

$$L^T g = \mathbf{0}$$

Sums up forces on all matching dofs

Coupling the physical model: 3-field form

Note: \mathbf{L} is also used to define a unique set of assembled/global dofs



$$\mathbf{L}^T \mathbf{g} = \begin{bmatrix} 1 & 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 & 1 \end{bmatrix} \begin{bmatrix} g_2^{(1)} \\ g_1^{(2)} \\ g_2^{(2)} \\ g_3^{(2)} \\ g_1^{(3)} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} \Leftrightarrow \begin{bmatrix} u_2^{(1)} \\ u_1^{(2)} \\ u_2^{(2)} \\ u_3^{(2)} \\ u_1^{(3)} \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} u_I \\ u_{III} \\ u_{III} \end{bmatrix} = \mathbf{L} \mathbf{u}_{glob}$$

and since global dofs are per definition compatible:

$$\mathbf{B} \mathbf{u}_{compatible} = \mathbf{B} \mathbf{L} \mathbf{u}_{glob} = \mathbf{0} \quad \forall \mathbf{u}_{glob} \quad \Rightarrow \quad \boxed{\mathbf{B} \mathbf{L} = \mathbf{0}} \\ \mathbf{L} = \text{null}(\mathbf{B})$$

Coupling the physical model: 3-field form

- Summarizing:

- an assembled system is described by

$$\begin{cases} M\ddot{u} + C\dot{u} + Ku = f + g \\ Bu = o \\ L^T g = o \end{cases}$$

→ local equilibrium

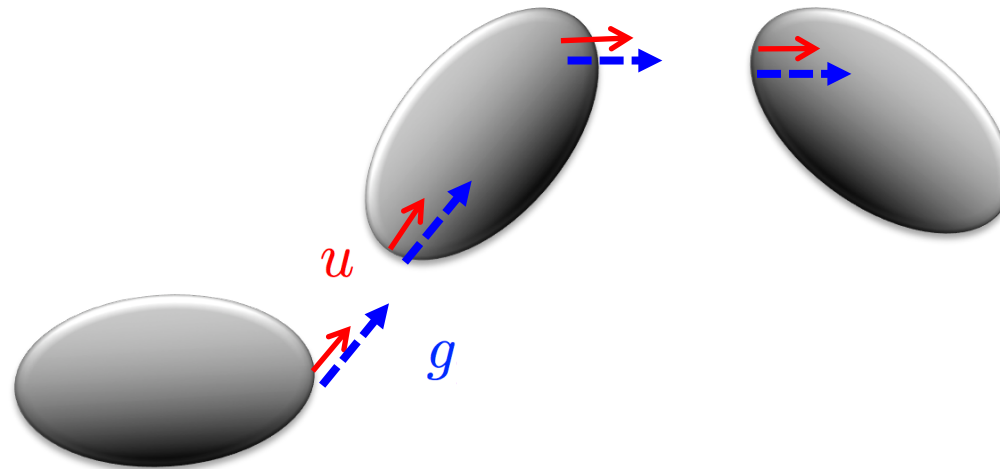
→ interface compatibility

→ equilibrium of connecting forces

3-field formulation, using block-matrices of unassembled quantities

- Additional property

$$BL = o$$



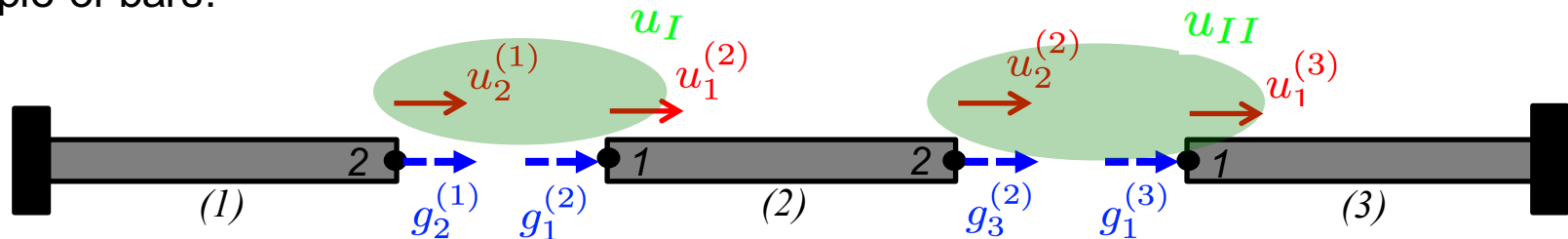
Coupling the physical model: **Primal assembly**

- Let us now write the assembled problem by considering from the start only a set of **dofs** that are compatible, i.e. originating from a **common/global assembled** set

$$\mathbf{u}_{glob} \quad \text{such that} \quad \mathbf{u} = \mathbf{L}\mathbf{u}_{glob}$$

In other words: the substructure $\mathbf{u}^{(s)}$ are picked from a unique global set \mathbf{u}_{glob}

- Example of bars.



$$\begin{bmatrix} u_2^{(1)} \\ u_1^{(2)} \\ u_2^{(2)} \\ u_1^{(3)} \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 1 & 0 \\ 0 & 1 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} u_I \\ u_{II} \end{bmatrix} = \mathbf{L}\mathbf{u}_{glob}$$

Coupling the physical model: **Primal assembly**

$$\mathbf{u}_{glob} \quad \text{such that} \quad \mathbf{u} = \mathbf{L}\mathbf{u}_{glob}$$

$$\left\{ \begin{array}{l} \mathbf{M}\ddot{\mathbf{u}} + \mathbf{C}\dot{\mathbf{u}} + \mathbf{K}\mathbf{u} = \mathbf{f} + \mathbf{g} \\ \mathbf{B}\mathbf{u} = \mathbf{0} \text{ naturally satisfied} \\ \mathbf{L}^T \mathbf{g} = \mathbf{0} \end{array} \right. \rightarrow \left\{ \begin{array}{l} \mathbf{M}\mathbf{L}\ddot{\mathbf{u}}_{glob} + \mathbf{C}\mathbf{L}\dot{\mathbf{u}}_{glob} + \mathbf{K}\mathbf{L}\mathbf{u}_{glob} = \mathbf{f} + \mathbf{g} \\ \mathbf{L}^T \mathbf{g} = \mathbf{0} \end{array} \right.$$

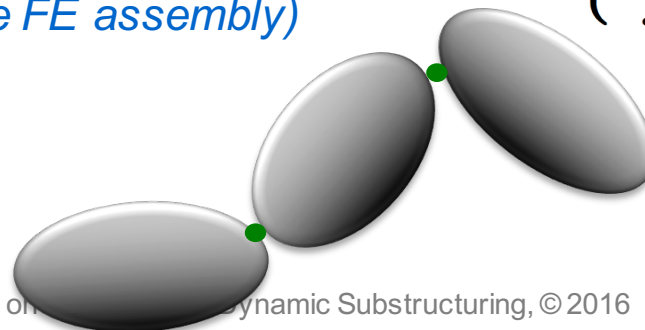
and one can eliminate the unknown interface forces
by assembling the local equilibriums and the interface:

$$\mathbf{L}^T \left(\mathbf{M}\mathbf{L}\ddot{\mathbf{u}}_{glob} + \mathbf{C}\mathbf{L}\dot{\mathbf{u}}_{glob} + \mathbf{K}\mathbf{L}\mathbf{u}_{glob} = \mathbf{f} + \cancel{\mathbf{g}} \right)$$

$$\tilde{\mathbf{M}}\ddot{\mathbf{u}}_{glob} + \tilde{\mathbf{C}}\dot{\mathbf{u}}_{glob} + \tilde{\mathbf{K}}\mathbf{u}_{glob} = \tilde{\mathbf{f}}$$

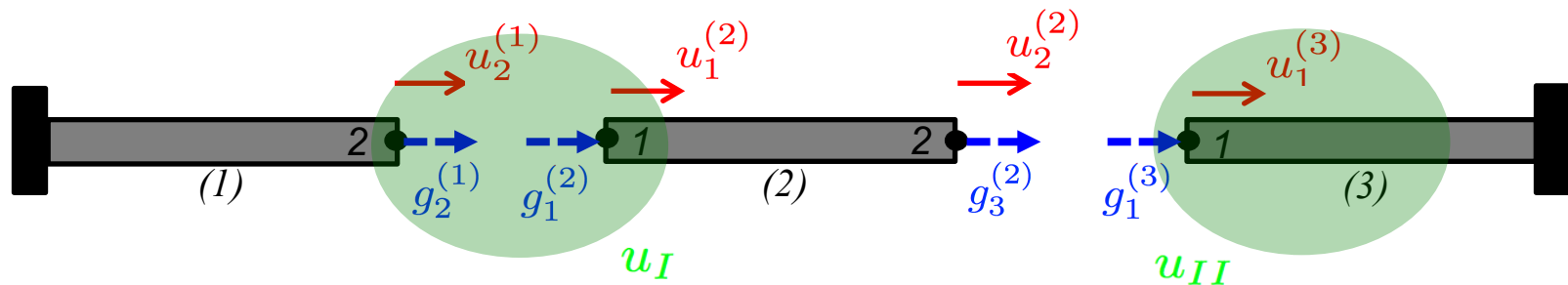
Primally assembled (like FE assembly)

$$\left\{ \begin{array}{ll} \tilde{\mathbf{M}} & \triangleq \mathbf{L}^T \mathbf{M} \mathbf{L} \\ \tilde{\mathbf{C}} & \triangleq \mathbf{L}^T \mathbf{C} \mathbf{L} \\ \tilde{\mathbf{K}} & \triangleq \mathbf{L}^T \mathbf{K} \mathbf{L} \\ \tilde{\mathbf{f}} & \triangleq \mathbf{L}^T \mathbf{f} \end{array} \right.$$



Coupling the physical model: Primal assembly

- Example of bars.



$$\tilde{\mathbf{K}} \triangleq \mathbf{L}^T \mathbf{K} \mathbf{L} = \begin{bmatrix} 1 & 1 & 0 & 0 \\ 0 & 0 & 1 & 1 \end{bmatrix} \begin{bmatrix} \boxed{k^{(1)}} & 0 & 0 & 0 \\ 0 & \boxed{k^{(2)}} & \boxed{-k^{(2)}} & 0 \\ 0 & \boxed{-k^{(2)}} & \boxed{k^{(2)}} & 0 \\ 0 & 0 & 0 & \boxed{k^{(3)}} \end{bmatrix} \begin{bmatrix} 1 & 0 \\ 1 & 0 \\ 0 & 1 \\ 0 & 1 \end{bmatrix} = \begin{bmatrix} \boxed{k^{(1)} + k^{(2)}} & \boxed{-k^{(2)}} \\ \boxed{-k^{(2)}} & \boxed{k^{(2)} + k^{(3)}} \end{bmatrix}$$

exactly like in Finite Elements.

Coupling the physical model: Dual assembly

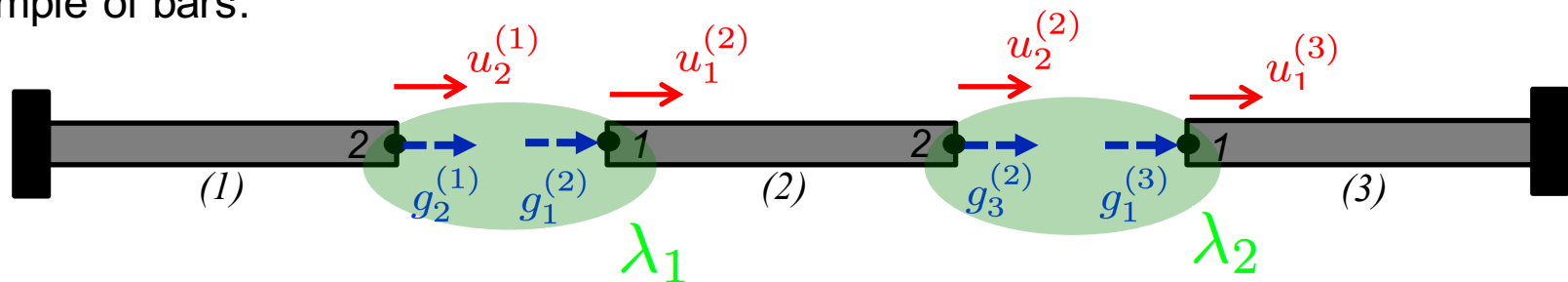
- Starting again from the 3-field formulation, let us now write the assembled problem by considering from the start only a set of **interface forces** that are in equilibrium, i.e. **equal an opposite**:

$$\begin{cases} M\ddot{u} + C\dot{u} + Ku = f + g \\ Bu = 0 \\ L^T g = 0 \end{cases}$$

$$\lambda \text{ such that } g = -B^T \lambda$$

In other words: the interface forces are $\pm \lambda$ on each side of the interface

- Example of bars.



$$g = \begin{bmatrix} 1 & 0 \\ -1 & 0 \\ 0 & 1 \\ 0 & -1 \end{bmatrix} \begin{bmatrix} \lambda_1 \\ \lambda_2 \end{bmatrix} = B^T \lambda$$

Coupling the physical model: Dual assembly

$$\lambda \text{ such that } g = -B^T \lambda$$

$$\begin{cases} M\ddot{u} + C\dot{u} + Ku = f + g \\ Bu = 0 \\ \textcircled{L^T g = 0} \text{ naturally satisfied} \end{cases}$$



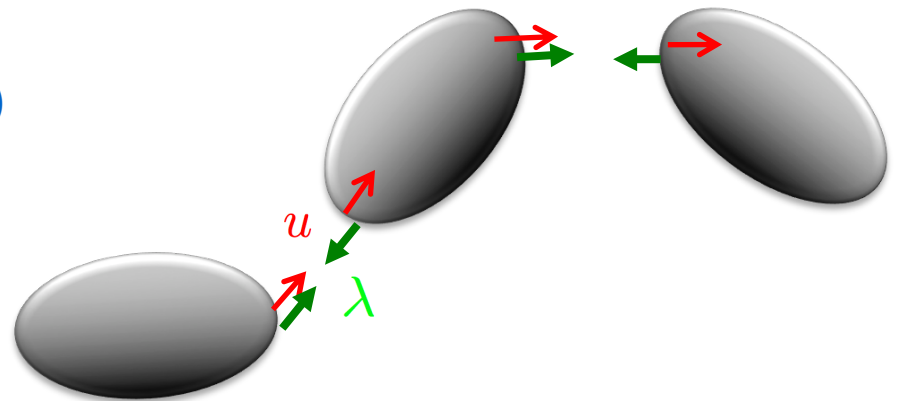
$$\begin{cases} M\ddot{u} + C\dot{u} + Ku + B^T \lambda = f \\ Bu = 0 \end{cases}$$

and in block matrix form

$$\begin{bmatrix} M & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \ddot{u} \\ \lambda \end{bmatrix} + \begin{bmatrix} C & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \dot{u} \\ \lambda \end{bmatrix} + \begin{bmatrix} K & B \\ B^T & 0 \end{bmatrix} \begin{bmatrix} u \\ \lambda \end{bmatrix} = \begin{bmatrix} f \\ 0 \end{bmatrix}$$

Dually assembled

*(like in constrained multibody formulation,
 λ = Lagrange multiplier)*

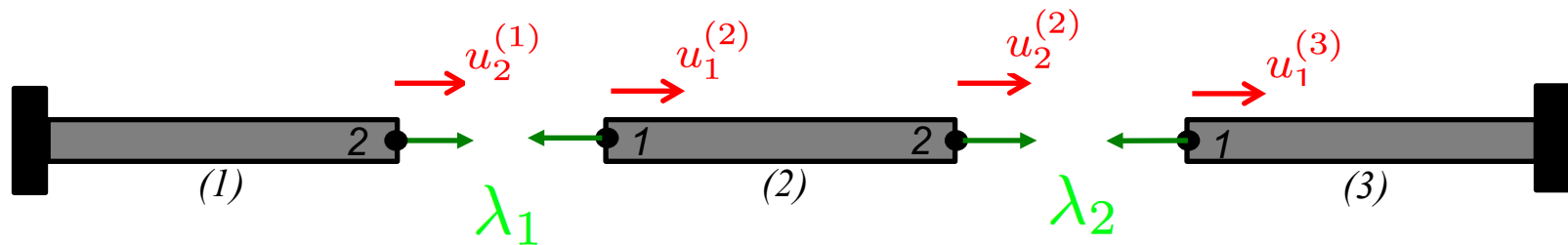


Note:

The compatibility condition could also be written on the velocities or accelerations. Careful for drift.

Coupling the physical model: Dual assembly

- Example of bars.



$$\begin{bmatrix} \mathbf{K} & \mathbf{B} \\ \mathbf{B}^T & \mathbf{o} \end{bmatrix} \begin{bmatrix} \mathbf{u} \\ \boldsymbol{\lambda} \end{bmatrix} = \begin{bmatrix} \boxed{k^{(1)}} & 0 & 0 & 0 \\ 0 & \boxed{k^{(2)}} & \boxed{-k^{(2)}} & 0 \\ 0 & \boxed{-k^{(2)}} & \boxed{k^{(2)}} & 0 \\ 0 & 0 & 0 & \boxed{k^{(3)}} \\ \hline 1 & -1 & 0 & 0 \\ 0 & 0 & 1 & -1 \end{bmatrix} \begin{bmatrix} \begin{bmatrix} 1 & 0 \\ -1 & 0 \\ 0 & 1 \\ 0 & -1 \end{bmatrix} \\ \mathbf{o} \end{bmatrix} \begin{bmatrix} \boxed{u_2^{(1)}} \\ \boxed{u_1^{(2)}} \\ \boxed{u_2^{(2)}} \\ \boxed{u_1^{(3)}} \\ \hline \lambda_1 \\ \lambda_2 \end{bmatrix}$$

Coupling the physical model: **SUMMARY**

$$\begin{cases} M\ddot{u} + C\dot{u} + Ku = f + g \\ Bu = 0 \\ L^T g = 0 \end{cases}$$

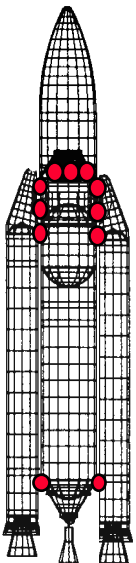
3-field

All mathematically equivalent !

$$\tilde{M}\ddot{u}_{glob} + \tilde{C}\dot{u}_{glob} + \tilde{K}u_{glob} = \tilde{f}$$

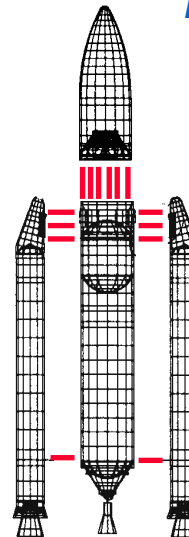
$$\begin{bmatrix} M & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \ddot{u} \\ \lambda \end{bmatrix} + \begin{bmatrix} C & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \dot{u} \\ \lambda \end{bmatrix} + \begin{bmatrix} K & B \\ B^T & 0 \end{bmatrix} \begin{bmatrix} u \\ \lambda \end{bmatrix} = \begin{bmatrix} f \\ 0 \end{bmatrix}$$

Primal



$$\begin{bmatrix} K^{(1)} & & 0 \\ & \ddots & \\ 0 & & K^{(N)} \end{bmatrix} \begin{bmatrix} u^{(1)} \\ \vdots \\ u^{(N)} \end{bmatrix} = \begin{bmatrix} f^{(1)} \\ \vdots \\ f^{(N)} \end{bmatrix}$$

Dual



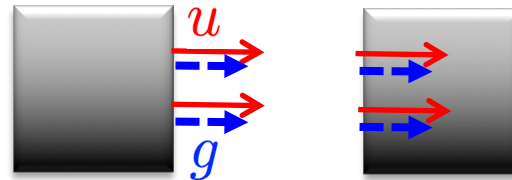
$$\begin{bmatrix} K^{(1)} & & 0 & B^{(1)T} \\ & \ddots & & \\ 0 & & K^{(N)} & B^{(N)T} \\ B^{(1)} & \dots & B^{(N)} & 0 \end{bmatrix} \begin{bmatrix} u^{(1)} \\ \vdots \\ u^{(N)} \\ \lambda \end{bmatrix} = \begin{bmatrix} f^{(1)} \\ \vdots \\ f^{(N)} \\ 0 \end{bmatrix}$$

Coupling the physical model: REMARKS

- When enforcing interface equilibrium and compatibility, we have assumed that the interface has no local dynamics:

$$Bu = \mathbf{0}$$

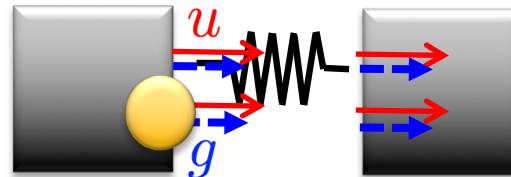
$$L^T g = \mathbf{0}$$



If “simple” interface flexibility, damping or mass is present at the interface, then the interface conditions must be slightly altered

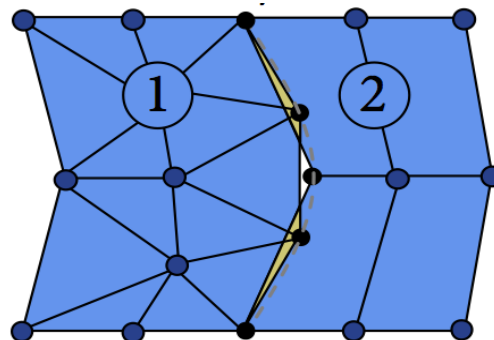
$$Bu = \cancel{\mathbf{0}}$$

$$L^T g = \cancel{\mathbf{0}}$$



and modified primal and dual assembled systems can be derived [2,3]

- If incompatible interface, the dual assembly can still be applied (B non Boolean) e.g. [5]



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 - ❑ Assembled admittance
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 - ❑ By filtering
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References and bibliography

Coupling in Frequency Domain

- Exactly the same assembly procedures

$$\begin{cases} M\ddot{u} + C\dot{u} + Ku = f + g \\ Bu = \mathbf{o} \\ L^T g = \mathbf{o} \end{cases}$$

$$\begin{cases} Z(\omega)u(j\omega) = f(\omega) + g(\omega) \\ Bu(\omega) = \mathbf{o} \\ L^T g(\omega) = \mathbf{o} \end{cases}$$

3-field

$$Z(j\omega) \triangleq -\omega^2 M + j\omega C + K$$

Structural **impedance** matrix

$$\tilde{Z}q = \tilde{f}$$

Primal

With the assembled impedance

$$\tilde{Z} \triangleq L^T Z L$$

$$\begin{bmatrix} Z & B^T \\ B & \mathbf{o} \end{bmatrix} \begin{bmatrix} u \\ \lambda \end{bmatrix} = \begin{bmatrix} f \\ \mathbf{o} \end{bmatrix}$$

where Z is the block-matrix of impedances of each individual substructure (i.e. free interface)

Commonly called

Frequency Based Substructuring or FBS

Coupling in Frequency Domain: Assembled Admittance

The dual assembly is usually preferred when the substructure data are coming from measurements. Why ?

$$\begin{bmatrix} \mathbf{Z} & \mathbf{B}^T \\ \mathbf{B} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \mathbf{u} \\ \lambda \end{bmatrix} = \begin{bmatrix} \mathbf{f} \\ \mathbf{0} \end{bmatrix} \rightarrow \mathbf{u} = \mathbf{Y} \left(\mathbf{f} - \mathbf{B}^T \lambda \right)$$

where $\mathbf{Y}(j\omega) \triangleq \mathbf{Z}^{-1} = \left(-\omega^2 \mathbf{M} + j\omega \mathbf{C} + \mathbf{K} \right)$

*Block-matrix of substructure **admittances***

$$\mathbf{B} \left(\mathbf{Y} \mathbf{f} - \mathbf{Y} \mathbf{B}^T \lambda \right) = \mathbf{0}$$

$$(\mathbf{B} \mathbf{Y} \mathbf{B}) \lambda = \mathbf{B} \mathbf{Y} \mathbf{f}$$

Dual interface problem:

“how much interface force is needed to close the interface gap resulting from external forces”

$$\begin{aligned} \mathbf{u} &= \mathbf{Y} \mathbf{f} - \mathbf{Y} \mathbf{B}^T (\mathbf{B} \mathbf{Y} \mathbf{B}^T)^{-1} \mathbf{B} \mathbf{Y} \mathbf{f} \\ &= \underbrace{(\mathbf{Y} - \mathbf{Y} \mathbf{B}^T (\mathbf{B} \mathbf{Y} \mathbf{B}^T)^{-1} \mathbf{B} \mathbf{Y})}_{\text{=Assembled Admittance}} \mathbf{f} \end{aligned}$$

=Assembled Admittance

Coupling in Frequency Domain: Assembled Admittance

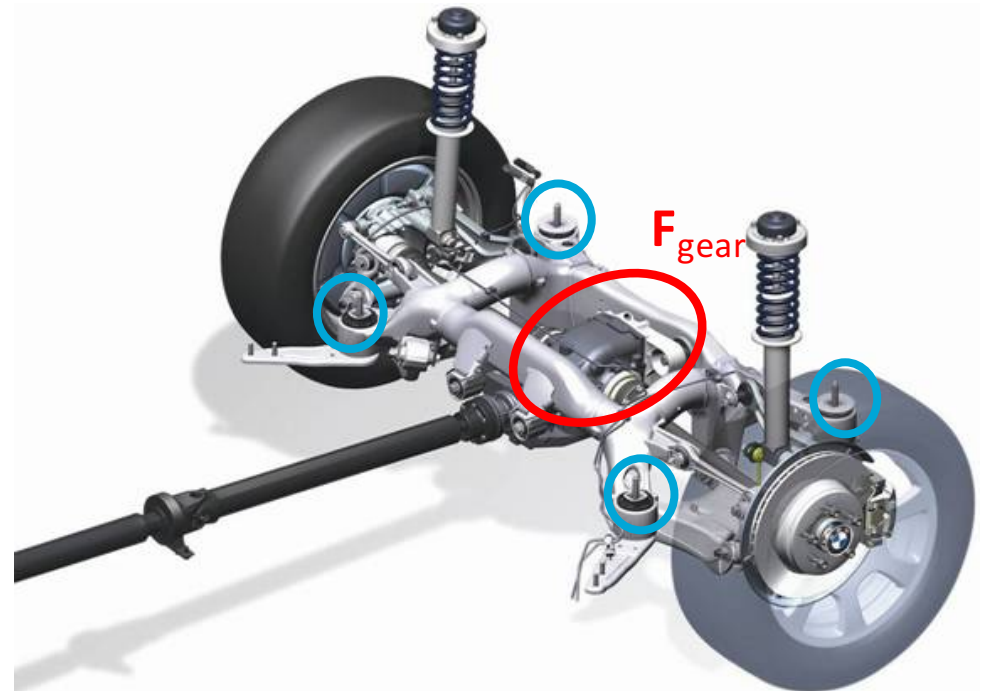
Summarizing:

If one knows the admittance of each substructure free on its interface, then the dynamic response of the assembled system is

$$u = (Y - YB^T(BYB^T)^{-1}BY)f$$

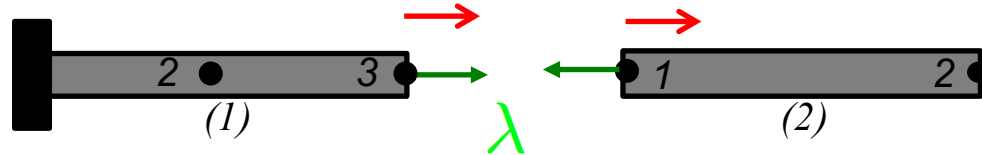
i.e. one can compute the admittance of the assembled system using local admittances !

→ useful in experimental dynamics
where admittances of non-connected
components are measured.



Coupling in Frequency Domain: Example

- Example of bars.



$$u = (Y - YB^T(BYB^T)^{-1}BY)f$$

$$Y = \begin{bmatrix} Y_{22}^{(1)} & Y_{23}^{(1)} & 0 & 0 \\ Y_{32}^{(1)} & Y_{33}^{(1)} & 0 & 0 \\ 0 & 0 & Y_{11}^{(2)} & Y_{12}^{(2)} \\ 0 & 0 & Y_{21}^{(2)} & Y_{22}^{(2)} \end{bmatrix}$$

$$B = \begin{bmatrix} 0 & 1 & -1 & 0 \end{bmatrix}$$

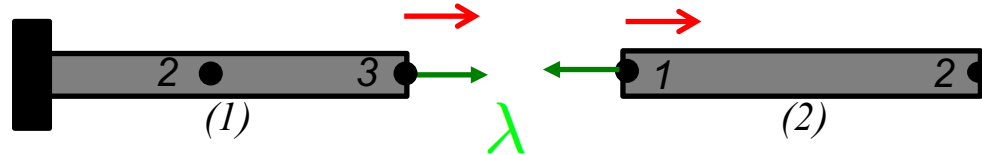
$$u = \begin{bmatrix} u_2^{(1)} \\ u_3^{(1)} \\ u_1^{(2)} \\ u_2^{(2)} \end{bmatrix}$$

$$\begin{bmatrix} u_2^{(1)} \\ u_3^{(1)} \\ u_1^{(2)} \\ u_2^{(2)} \end{bmatrix} = \left\{ \begin{bmatrix} Y_{22}^{(1)} & Y_{23}^{(1)} & 0 & 0 \\ Y_{32}^{(1)} & Y_{33}^{(1)} & 0 & 0 \\ 0 & 0 & Y_{11}^{(2)} & Y_{12}^{(2)} \\ 0 & 0 & Y_{21}^{(2)} & Y_{22}^{(2)} \end{bmatrix} - \begin{bmatrix} Y_{23}^{(1)} \\ Y_{33}^{(1)} \\ -Y_{11}^{(2)} \\ -Y_{21}^{(2)} \end{bmatrix} \left(Y_{33}^{(1)} + Y_{11}^{(2)} \right)^{-1} \begin{bmatrix} Y_{32}^{(1)} & Y_{33}^{(1)} & -Y_{11}^{(2)} & -Y_{12}^{(2)} \end{bmatrix} \right\} \begin{bmatrix} f_2^{(1)} \\ f_3^{(1)} \\ f_1^{(2)} \\ f_2^{(2)} \end{bmatrix}$$

assembled interface flexibility

Coupling in Frequency Domain: Example

- Example of bars.



$$u = (Y - YB^T(BYB^T)^{-1}BY)f$$

$$Y = \begin{bmatrix} Y_{22}^{(1)} & Y_{23}^{(1)} & 0 & 0 \\ Y_{32}^{(1)} & Y_{33}^{(1)} & 0 & 0 \\ 0 & 0 & Y_{11}^{(2)} & Y_{12}^{(2)} \\ 0 & 0 & Y_{21}^{(2)} & Y_{22}^{(2)} \end{bmatrix}$$

$$B = \begin{bmatrix} 0 & 1 & -1 & 0 \end{bmatrix}$$

$$u = \begin{bmatrix} u_2^{(1)} \\ u_3^{(1)} \\ u_1^{(2)} \\ u_2^{(2)} \end{bmatrix}$$

$$\begin{bmatrix} u_2^{(1)} \\ u_3^{(1)} \\ u_1^{(2)} \\ u_2^{(2)} \end{bmatrix} = \left\{ \begin{bmatrix} Y_{22}^{(1)} & Y_{23}^{(1)} & 0 & 0 \\ Y_{32}^{(1)} & Y_{33}^{(1)} & 0 & 0 \\ 0 & 0 & Y_{11}^{(2)} & Y_{12}^{(2)} \\ 0 & 0 & Y_{21}^{(2)} & Y_{22}^{(2)} \end{bmatrix} - \begin{bmatrix} Y_{23}^{(1)} \\ Y_{33}^{(1)} \\ -Y_{11}^{(2)} \\ -Y_{21}^{(2)} \end{bmatrix} \left(Y_{33}^{(1)} + Y_{11}^{(2)} \right)^{-1} \begin{bmatrix} Y_{32}^{(1)} & Y_{33}^{(1)} & -Y_{11}^{(2)} & -Y_{12}^{(2)} \end{bmatrix} \right\} \begin{bmatrix} f_2^{(1)} \\ f_3^{(1)} \\ f_1^{(2)} \\ f_2^{(2)} \end{bmatrix}$$

incompatibility between substructures when not connected

interface force λ needed for compatibility

response to applied forces

response to interface forces

Coupling in Frequency Domain: Summary

Summary to remember !

$$u = (Y - YB^T(BYB^T)^{-1}BY)f$$

where Y contains in its blocks the uncoupled admittances

where B is signed Boolean and describes the connection topology

$$u = (Y - YB^T(BYB^T)^{-1}BY)f$$

incompatibility between substructures
when no connection

interface force λ needed for compatibility

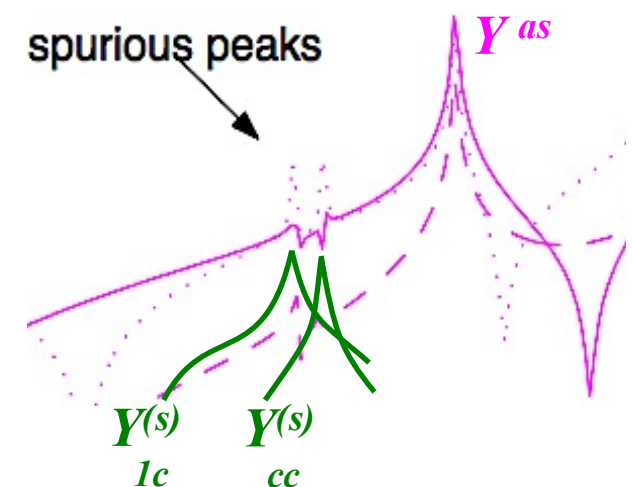
response
to applied forces

response
to interface forces

Coupling in Frequency Domain: Remarks

$$u = (Y - YB^T(BYB^T)^{-1}BY)f$$

- + Systematic assembly for any number of substructures and any topology
- - needs admittances for all interface dofs ...
 - - not always possible to place sensors on interface
 - - not always possible to measure **all** dofs on interface
- - implies *inverting* the assembled interface flexibility
→ explosion of measurement errors [6]
e.g. measurement noise
e.g. poles in $Y^{(s)}$ not identical for all coefficient [7]



Coupling in Frequency Domain: Remarks

$$u = (Y - YB^T(BYB^T)^{-1}BY)f$$

- + Systematic assembly for any number of substructures and any topology
 - - needs admittances for all interface dofs ...
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 - - not always possible to measure **all** dofs on interface
 - - implies *inverting* the assembled interface flexibility
→ explosion of measurement errors [6]
e.g. measurement noise
e.g. poles in $Y^{(s)}$ not identical for all coefficient [7]
- expansion of measurements in the vicinity of interface using local shapes (e.g. Serep). See an application in MOD08.
- SVD filtering of data, e.g. [8,13]
Modal synthesis of data (if low modal density)
Correction with FE model [9]
Cleaning Y^{as} *a posteriori* [10]
Weakening of interface compatibility → next sections

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Dual coupling in representation space (Modal)

- How do the assembled equations look if

each substructure is approximated by a
reduced set of (modal) shapes stored in the columns of $R^{(s)}$

$$u^{(s)} \simeq R^{(s)} \eta^{(s)} \quad \text{in local equilibria}$$

$$M^{(s)} \ddot{u}^{(s)}(t) + C \dot{u}^{(s)}(t) + K^{(s)} u^{(s)}(t) = f^{(s)}(t) + g^{(s)}(t)$$

$$M^{(s)} R^{(s)} \ddot{\eta}^{(s)} + C^{(s)} R^{(s)} \dot{\eta}^{(s)} + K^{(s)} R^{(s)} \eta^{(s)} = f^{(s)} + g^{(s)} + r^{(s)}$$

or in block notations

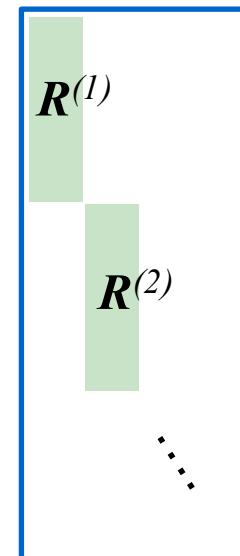
Equilibrium error due to response approximation

$$u \simeq R \eta$$

$$R \triangleq \text{diag} \left(R^{(1)}, \dots, R^{(n)} \right) \quad \text{in local equilibria}$$

$$M \ddot{u} + C \dot{u} + K u = f + g$$

$$M R \ddot{\eta} + C R \dot{\eta} + K R \eta = f + g + r$$



Dual coupling in representation space (Modal)

$$u \simeq R\eta \quad MR\ddot{\eta} + CR\dot{\eta} + KR\eta = f + g + r$$

Equilibrium error due to response approximation

Asking that the residual force creates no work in the approximation space (as done in reduction techniques – virtual work),

$$R^T (MR\ddot{\eta} + CR\dot{\eta} + KR\eta = f + g + \cancel{r})$$

→
$$M_m\ddot{\eta} + C_m\dot{\eta} + K_m\eta = f_m + g_m$$

where $\left\{ \begin{array}{l} M_m \triangleq R^T M R \\ C_m \triangleq R^T C R \\ K_m \triangleq R^T K R \\ f_m \triangleq R^T f \\ g_m \triangleq R^T g \end{array} \right.$ are the matrices in the representation space (reduced)

The compatibility condition $Bu = \mathbf{0}$ transforms into $B_m\eta = \mathbf{0}$

where $B_m \triangleq BR$

Dual coupling in representation space (Modal)

Following the same reasoning as for the general case, one can write the dually assembled problem in the representation space [1]

$$\begin{bmatrix} \mathbf{M}_m & \mathbf{0} \\ \mathbf{0} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\eta} \\ \lambda \end{bmatrix} + \begin{bmatrix} \mathbf{C}_m & \mathbf{0} \\ \mathbf{0} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \dot{\eta} \\ \lambda \end{bmatrix} + \begin{bmatrix} \mathbf{K}_m & \mathbf{B}_m \\ \mathbf{B}_m^T & \mathbf{0} \end{bmatrix} \begin{bmatrix} \eta \\ \lambda \end{bmatrix} = \begin{bmatrix} \mathbf{f}_m \\ \mathbf{0} \end{bmatrix}$$

REMARKS:

- When expressed in the frequency domain, the assembled problem in representation space is

$$\begin{bmatrix} \mathbf{Z}_m & \mathbf{B}_m^T \\ \mathbf{B}_m & \mathbf{0} \end{bmatrix} \begin{bmatrix} \eta \\ \lambda \end{bmatrix} = \begin{bmatrix} \mathbf{f}_m \\ \mathbf{0} \end{bmatrix} \quad \text{where} \quad \mathbf{Z}_m = \mathbf{R}^T \mathbf{Z} \mathbf{R}$$

- Looks very similar to the Dual problem in the physical space but ...
 - ... the matrices are reduced and expressing the physical properties in the representation space
 - ... the unknowns are now the amplitudes of the representation vectors
- It is important to build the reduction space \mathbf{R} on shapes that can properly represent the response of the assembled system
 - some methods use modes of the system when assembled with a dummy mimicking roughly neighboring structures (see the transmission simulator idea in MOD9)

Dual coupling in representation space (Modal)

$$\begin{bmatrix} \mathbf{M}_m & \mathbf{0} \\ \mathbf{0} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\eta} \\ \lambda \end{bmatrix} + \begin{bmatrix} \mathbf{C}_m & \mathbf{0} \\ \mathbf{0} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \dot{\eta} \\ \lambda \end{bmatrix} + \begin{bmatrix} \mathbf{K}_m & \mathbf{B}_m \\ \mathbf{B}_m^T & \mathbf{0} \end{bmatrix} \begin{bmatrix} \eta \\ \lambda \end{bmatrix} = \begin{bmatrix} \mathbf{f}_m \\ \mathbf{0} \end{bmatrix}$$

REMARKS (continued):

- If the approximation space is the modal space of the substructures (free interface),

$$\mathbf{u}^{(s)} = \boldsymbol{\Phi}^{(s)} \eta^{(s)} \quad \text{where} \quad \mathbf{K}^{(s)} \boldsymbol{\Phi}^{(s)} = \mathbf{M}^{(s)} \boldsymbol{\Phi}^{(s)} \boldsymbol{\Omega}^{(s)^2}$$

then in frequency domain (with proportional damping)

$$\begin{bmatrix} \begin{bmatrix} -\omega^2 \mathbf{I} + i\omega 2\zeta^{(1)} \boldsymbol{\Omega}^{(1)} + \boldsymbol{\Omega}^{(1)^2} & \mathbf{0} \\ \mathbf{0} & \ddots \end{bmatrix} & \begin{bmatrix} \boldsymbol{\Phi}^{(1)T} \mathbf{B}^{(1)T} \\ \vdots \end{bmatrix} \\ \begin{bmatrix} \mathbf{B}^{(1)} \boldsymbol{\Phi}^{(1)} & \dots \end{bmatrix} & \begin{bmatrix} \mathbf{0} \end{bmatrix} \end{bmatrix} \begin{bmatrix} \eta^{(1)} \\ \vdots \\ \lambda \end{bmatrix} = \begin{bmatrix} \boldsymbol{\Phi}^{(1)T} \mathbf{f}^{(1)} \\ \vdots \\ \mathbf{0} \end{bmatrix}$$

- One major issue: the condition $\mathbf{B} \mathbf{R} \eta = \mathbf{B}_m \eta = \mathbf{0}$ requires full compatibility on the interface. Since each substructures is represented by a small numbers of local modes, this might lead to **locking** (the compatibility can be satisfied only when 0 response)
→ need for weakening the interface compatibility

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Weak interface compatibility

Enforcing the strong compatibility for the entire interface can lead to serious problems when (see earlier discussions)

- the admittance contains measurement errors and the interface is stiff
- when the substructures are approximated by a limited number of representative shapes that do not necessarily match on the interface (locking)



2 different techniques to relax the interface compatibility (***weakening***)

1. requiring that only a projection of the interface gap is 0
→ **Projected compatibility**
2. projecting the interface displacements on each side of the interface on a common reduced space and enforcing compatibility only for that space
→ **Filtering**

Weak interface compatibility: Projected Compatibility

Starting from the frequency representation with physical dofs (would be similar for modal representation):

$$\begin{bmatrix} \mathbf{Z} & \mathbf{B}^T \\ \mathbf{B} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \mathbf{u} \\ \lambda \end{bmatrix} = \begin{bmatrix} \mathbf{f} \\ \mathbf{0} \end{bmatrix}$$

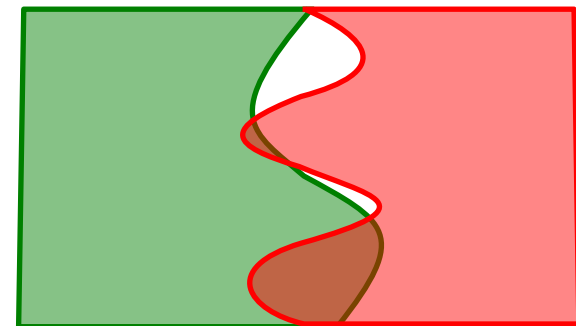
In order to weaken the compatibility condition, we will require only that

the projection of the interface gap on a small number of interface modes is 0

$$\boldsymbol{\Gamma}_I^T (\mathbf{B}\mathbf{u}) = 0$$

where $\boldsymbol{\Gamma}_I$ contains the interface projection shapes.

For instance if $\boldsymbol{\Gamma}_I = \begin{bmatrix} 1 \\ 1 \\ \vdots \\ 1 \end{bmatrix}, \dots$



..... one imposes only an average compatibility

Weak interface compatibility: Projected Compatibility

One can show that the dual problem then writes

$$\begin{bmatrix} \mathbf{Z} & \mathbf{B}^T \mathbf{\Gamma}_I \\ \mathbf{\Gamma}_I^T \mathbf{B} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \mathbf{u} \\ \lambda_\Gamma \end{bmatrix} = \begin{bmatrix} \mathbf{f} \\ \mathbf{0} \end{bmatrix}$$

- small number of compatibility conditions on the interface (projected)
- small number of interface force variables λ_Γ
- can be interpreted as if the compatibility is enforced through interface forces that are built on the interface projection shapes: $\lambda \simeq \mathbf{\Gamma}_I \lambda_\Gamma$
- Similarly, if the substructure displacements are approximated in a representation space:

$$\mathbf{u} \simeq \mathbf{R}\eta \quad \text{and} \quad \begin{bmatrix} \mathbf{Z}_m & \mathbf{B}_m^T \mathbf{\Gamma}_I \\ \mathbf{\Gamma}_I^T \mathbf{B}_m & \mathbf{0} \end{bmatrix} \begin{bmatrix} \eta \\ \lambda_\Gamma \end{bmatrix} = \begin{bmatrix} \mathbf{f}_m \\ \mathbf{0} \end{bmatrix}$$

in which the displacements **and** the interface forces are approximated.

Weak interface compatibility: Filtering

On the second approach, one fits the interface displacement on both sides of the interface on a common displacement shape.

Calling Φ_Γ the assumed modes for the interface displacements,

$$\mathbf{u}^{(s)} = \begin{bmatrix} \mathbf{u}_i^{(s)} \\ \mathbf{u}_c^{(s)} \end{bmatrix} = \begin{bmatrix} \mathbf{u}_i^{(s)} \\ \Phi_\Gamma \alpha^{(s)} + \tilde{\mathbf{u}}_c^{(s)} \end{bmatrix}$$

→ internal dofs
→ connecting dofs
↘ part not representable in Φ_Γ
↙ common interface modes

One will enforce a weak compatibility by asking only that

the amplitudes of Φ_Γ should be equal on each side of the interface: $B\alpha = 0$

$$\mathbf{u}_c^{(s)} \simeq \Phi_\Gamma \alpha^{(s)}$$

and by least square

$$\alpha^{(s)} = (\Phi_\Gamma^T \Phi_\Gamma)^{-1} \Phi_\Gamma^T \mathbf{u}_c^{(s)}$$

$$\Rightarrow B(\Phi_\Gamma^T \Phi_\Gamma)^{-1} \Phi_\Gamma^T \mathbf{u}_c = BF\mathbf{u} = 0$$

↙ filter that removes everything that is not in the space of Φ_Γ

Weak interface compatibility: Filtering

Finally the dual problem writes

$$\begin{bmatrix} \mathbf{Z} & \mathbf{F}^T \mathbf{B}^T \\ \mathbf{B} \mathbf{F} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \mathbf{u} \\ \lambda_\alpha \end{bmatrix} = \begin{bmatrix} \mathbf{f} \\ \mathbf{0} \end{bmatrix}$$

- similarly if the displacements are approximated in a representation space

$$\begin{bmatrix} \mathbf{Z}_m & \mathbf{R}^T \mathbf{F}^T \mathbf{B}^T \\ \mathbf{B} \mathbf{F} \mathbf{R} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \eta \\ \lambda_\alpha \end{bmatrix} = \begin{bmatrix} \mathbf{f}_m \\ \mathbf{0} \end{bmatrix}$$

- λ_α has the meaning of generalized forces associated to the interface displacements
- An important application: when Φ_I contains the rigid body motion of the interface, then λ_α are the 6 forces and moments resultants at the interface
 - often used for rigid interfaces (bolts, brackets), allowing to construct rotational dofs
 - in that case the method is also called EMPC (Explicit Multi-Point connection) or virtual point transformation [11,12]

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Summary

$$\begin{bmatrix} \mathbf{Z} & \mathbf{B}^T \\ \mathbf{B} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \mathbf{u} \\ \lambda \end{bmatrix} = \begin{bmatrix} \mathbf{f} \\ \mathbf{0} \end{bmatrix} \quad \Leftrightarrow \quad \overbrace{u = (Y - YB^T(BYB^T)^{-1}BY)f}^{\text{assembled admittance}}$$

or

$$\begin{bmatrix} \mathbf{Z}_m & \mathbf{B}_m^T \\ \mathbf{B}_m & \mathbf{0} \end{bmatrix} \begin{bmatrix} \eta \\ \lambda \end{bmatrix} = \begin{bmatrix} \mathbf{f}_m \\ \mathbf{0} \end{bmatrix} \quad \text{in reduced representation space } \mathbf{u} \simeq \mathbf{R}\eta$$

$$\mathbf{Z}_m = \mathbf{R}^T \mathbf{Z} \mathbf{R}$$

Weakening of compatibility by *projection*:

$$\lambda \simeq \Gamma_I \lambda_\Gamma$$

$$\begin{bmatrix} \mathbf{Z} & \mathbf{B}^T \Gamma_I \\ \Gamma_I^T \mathbf{B} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \mathbf{u} \\ \lambda_\Gamma \end{bmatrix} = \begin{bmatrix} \mathbf{f} \\ \mathbf{0} \end{bmatrix}$$

or

$$\begin{bmatrix} \mathbf{Z}_m & \mathbf{B}_m^T \Gamma_I \\ \Gamma_I^T \mathbf{B}_m & \mathbf{0} \end{bmatrix} \begin{bmatrix} \eta \\ \lambda_\Gamma \end{bmatrix} = \begin{bmatrix} \mathbf{f}_m \\ \mathbf{0} \end{bmatrix}$$

Weakening of compatibility by *filtering*:

$$\mathbf{u}_c^{(s)} \simeq \Phi_\Gamma \alpha^{(s)} \\ \mathbf{F} = (\Phi_\Gamma^T \Phi_\Gamma)^{-1} \Phi_\Gamma^T$$

$$\begin{bmatrix} \mathbf{Z} & \mathbf{F}^T \mathbf{B}^T \\ \mathbf{B} \mathbf{F} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \mathbf{u} \\ \lambda_\alpha \end{bmatrix} = \begin{bmatrix} \mathbf{f} \\ \mathbf{0} \end{bmatrix}$$

or

$$\begin{bmatrix} \mathbf{Z}_m & \mathbf{R}^T \mathbf{F}^T \mathbf{B}^T \\ \mathbf{B} \mathbf{F} \mathbf{R} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \eta \\ \lambda_\alpha \end{bmatrix} = \begin{bmatrix} \mathbf{f}_m \\ \mathbf{0} \end{bmatrix}$$

References and bibliography (*non exhaustive!*)

- 1 D. de Klerk, D. J. Rixen, and S. N. Voormeeren. General framework for dynamic substructuring: History, review and classification of techniques. *AIAA Journal*, 46(5):1169–1181, 2008.
- 2 S. N. Voormeeren, P. L. C. van der Valk, and D. J. Rixen. Generalized Methodology for Assembly and Reduction of Component Models for Dynamic Substructuring. *AIAA Journal*, 49:1010–1020, May 2011.
- 3 S. Voormeeren. *Dynamic Substructuring Methodologies for Integrated Dynamic Analysis of Wind Turbines*. PhD thesis, Delft University of Technology, 2012.
- 4 D. J. Rixen and P. L. van der Valk. An impulse based substructuring approach for impact analysis and load case simulations. *Journal of Sound and Vibration*, 332:7174–7190, 2013.
- 5 D. Rixen. *Substructuring and Dual Methods in Structural Analysis*. PhD thesis, Université de Liège, Belgium, Collection des Publications de la Faculté des Sciences appliquées, n° 175, 1997.
- 6 S. Voormeeren, D. de Klerk, and D. Rixen. Uncertainty quantification in experimental frequency based substructuring. *Mechanical Systems and Signal Processing*, 24(1):106 – 118, 2010.
- 7 D. J. Rixen. How measurement inaccuracies induce spurious peaks in frequency based substructuring. In *IMAC-XXVII: International Modal Analysis Conference, Orlando, FL*, Bethel, CT, February 2008. Society for Experimental Mechanics.
- 8 K. Sanliturk and O. Cakar. Noise elimination from measured frequency response functions. *Mechanical Systems and Signal Processing*, 19(3):615–631, 2005.
- 9 T. G. Carne and C. R. Dohrmann. Improving experimental frequency response function matrices for admittance modeling. In *IMAC-XXIV: International Modal Analysis Conference, St Louis, MO*, Bethel, CT, February 2006. Society for Experimental Mechanics.
- 10 D. Nicgorski and P. Avitabile. Conditioning of frf measurements for use with frequency based substructuring. *Mechanical Systems and Signal Processing*, 24(2):340–351, 2010.

References and bibliography (*non exhaustive!*)

- 11 *Solving the RDoF Problem in Experimental Dynamic Substructuring*, D. de Klerk, D.J. Rixen, S.N. Voormeeren and F. Pasteuning, International Modal Analysis Conference, IMAC-XXVI, SEM, Orlando, FL, 4-7 Feb., 2008.
- 12 M. van der Seijs, D. de Klerk, D. Rixen, and S. Rahimi. Validation of current state frequency based substructuring technology for the characterisation of steering gear – vehicle interaction. In *IMAC-XXXI: International Modal Analysis Conference, Garden Grove, California USA*, Bethel, CT, February Feb. 11-14, 2013. Society for Experimental Mechanics.
- 13 D. Otte, J. Leuridan, H. Grangier, and R. Aquilina. Prediction of the dynamics of structural assemblies using measured frf-data: some improved data enhancement techniques. In *IMAC-IX: International Modal Analysis Conference, Florence, Italy*, pages 909–918, Bethel, CT, February 1991. Society for Experimental Mechanics.